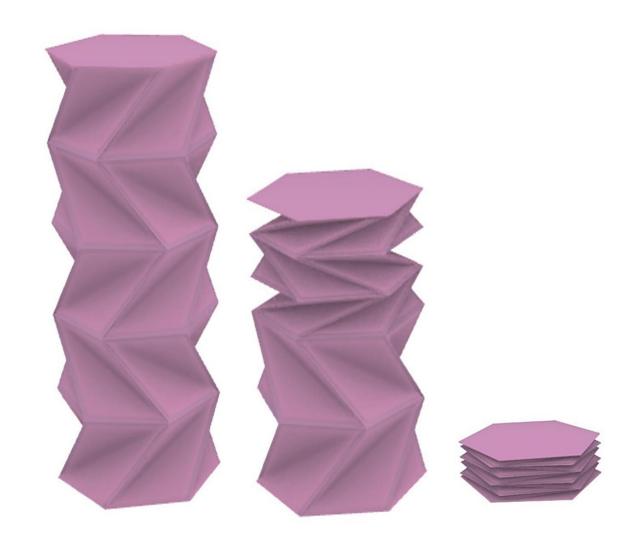
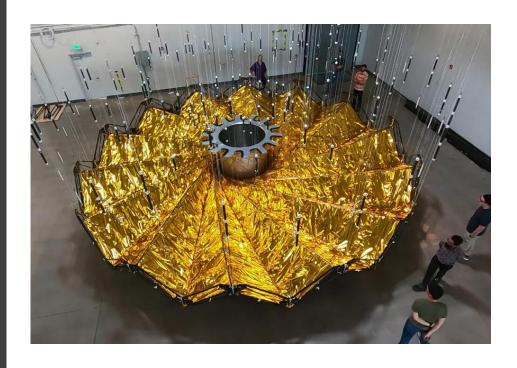
From Paper to Space: Testing Origami-Inspired Structures in a CubeSat Mission

Ines Uriol Balbin Ester Velázquez Navarro Pablo Solano Lopez





Origami-Inspired Structures





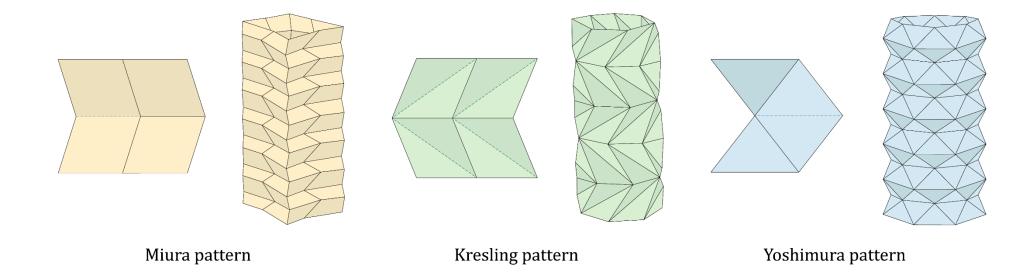


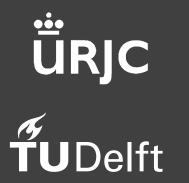
Credit: SAGA





Origami-Inspired Structures





Why a CubeSat Demonstrator?

- Small-scale proof-of-concept
- Explore the different technical solutions
- Explore manufacturing issues
- Raising the TRL







Requirements

- Design
 - Fit within 12U CubeSat volume
 - High Deployment Ratio, η = Deployed Volume/Folded Volume

- Structural Performance:
 - Withstand folding and deployment loads
 - Rigidity after deployment

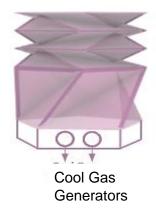




Design Solution

Kresling pattern made of dual-matrix composite with inflation deployment

- Lightweight
- Few mechanical parts
- Not always retractable

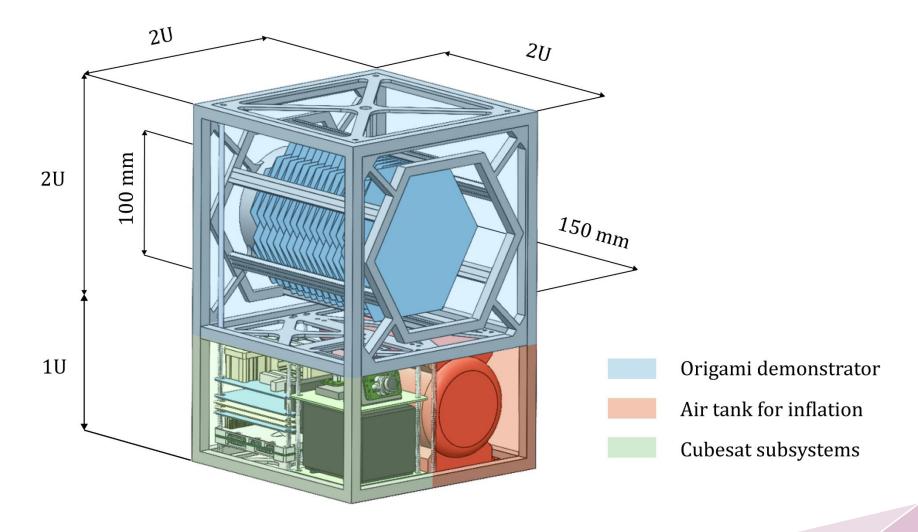


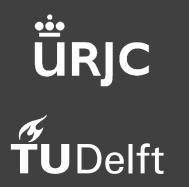






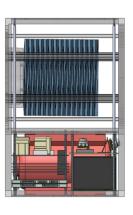
Inflation Deployment



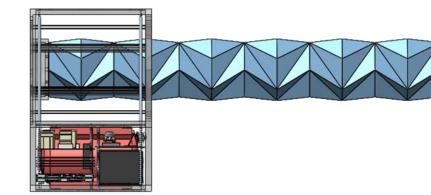




Inflation Deployment



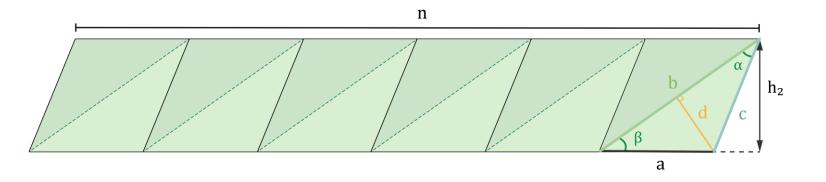
- Deployment by cool gas inflation
- Deployment ratio ≈ 13

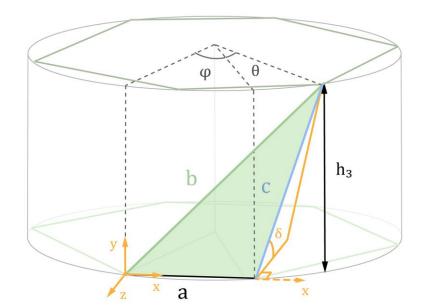


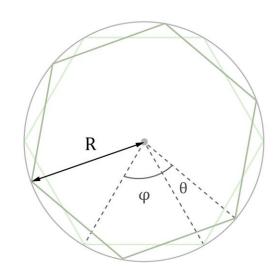


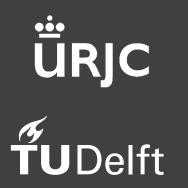


Kresling Pattern



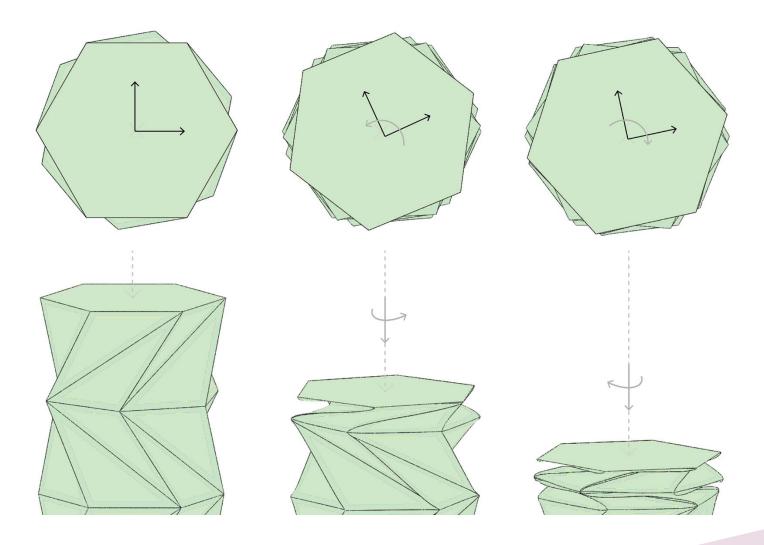


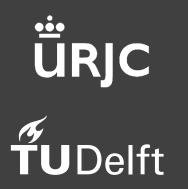






Kresling Pattern









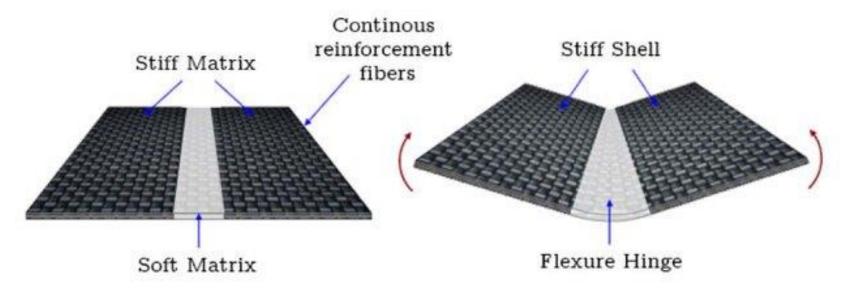








Dual-Matrix composite material

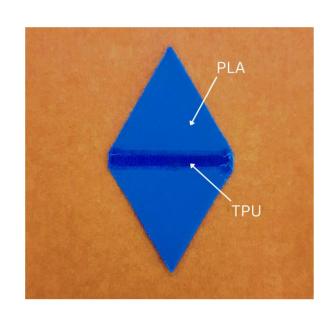


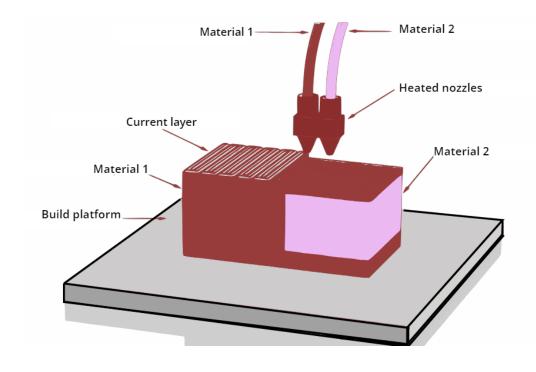






3D Printing with thermoplastic filaments





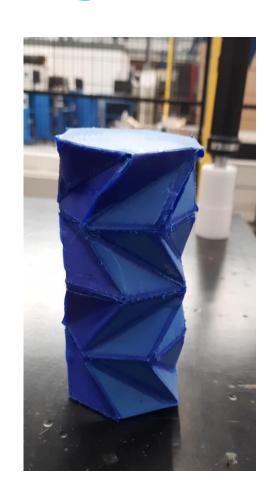




3D Printing with thermoplastic filaments











Conclusions and next steps

- Suitable design fit for a CubeSat mission
- Optimal configuration for Kresling pattern
- Suitable manufacturing and prototyping techniques
- Ongoing: Manufacturing and testing optimal configuration





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